Conformal Transformation and it's Applications

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Abstract

This paper investigates flow past a flat plate and two-dimensional irrotational motion of fluid due to singularities between two fixed boundaries. The flow past a flat plate is obtained by using Joukowski transformation. It is also shown by examples that the conformal transformation can make a problem of irrotational flow treatable by converting an awkwardly shaped boundary into one of the simple forms.

Keywords: flat plate, Joukowski transformation, conformal transformation, singularities.

1. Conformal Transformation

Suppose that z and ζ are two complex variables defined by z = x + iy and $\zeta = \xi + i\eta$ where x, y, ξ , η are real variables. Suppose that z describes a certain curve C in the z-plane and ζ is related to z by means of the transformation $\zeta = f(z)$ where f(z) is analytic.

If f(z) is a single-valued function of z, then to each point in the z-plane, we can obtain a corresponding point in the ζ -plane. In this way, the curve C in the z-plane may be mapped into a curve C' in the ζ -plane.



Figure 1

Suppose that the function f (z) is analytic. Let P, Q, R be neighboring points in the zplane such that OP = z, $OQ = z + \delta z_1$, $OR = z + \delta z_2$.



Under the transformation $\zeta = f(z)$, suppose that P,Q, R, map into the points P', Q',R' respectively in the ζ -plane, where OP' = ζ , OQ' = $\zeta + \delta \zeta_1$, OR'= $\zeta + \delta \zeta_2$. It is assumed that

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$$\begin{split} \left| \delta z_{1} \right|, \left| \delta z_{2} \right|, \left| \delta \zeta_{1} \right|, \left| \delta \zeta_{2} \right| & \text{are small. Since } f(z) \text{ is analytic, } \frac{d\zeta}{dz} \text{ is unique at P. Thus, to the first} \\ \text{order of smallness, } \frac{\delta \zeta_{1}}{\delta z_{1}} = \frac{\delta \zeta_{2}}{\delta z_{2}} \quad (\text{or)} \quad \frac{\delta \zeta_{1}}{\delta \zeta_{2}} = \frac{\delta z_{1}}{\delta z_{2}} \quad (1) \\ \text{Therefore, } \quad \left| \frac{\delta \zeta_{1}}{\delta \zeta_{2}} \right| = \left| \frac{\delta z_{1}}{\delta z_{2}} \right|, \quad (2) \quad \text{arg} \\ \delta \zeta_{1} - \arg \delta \zeta_{2} = \arg \delta z_{1} - \arg \delta z_{2}, \quad (3) \end{split}$$

and $\delta \zeta_1 = \frac{\delta \zeta_1}{\delta z_1} \, \delta \, z_1.$

Therefore, $\left|\delta\zeta_{1}\right| = \left|\frac{\delta\zeta_{1}}{\delta z_{1}}\right| \left|\delta z_{1}\right|$ and $\arg\delta\zeta_{1} = \arg\left(\frac{\delta\zeta_{1}}{\delta z_{1}}\right) + \arg\delta z_{1}$. So in the neighborhood of the

point P' distances are multiplied by the value of $\left|\frac{\delta \zeta_1}{\delta z_1}\right|$ at P; this is called the magnification of the transformation.

From (2) and (3), we obtain $\frac{P'Q'}{P'R'} = \frac{PQ}{PR}$ and $\angle R'P'Q' = \angle RPQ$. Thus the triangles R'P'Q' and RPQ are similar. So an infinitesimal triangle in the z-plane maps into a similar infinitesimal triangle in the ζ -plane. Thus the mapping preserves the angles and the similarity of corresponding infinitesimal triangles. Such a transformation which has these properties is said to be conformal.

Example: Transformation of $w = z^2$



Figure 3

1.1 Applications of Conformal Transformation

Suppose there is a two-dimensional incompressible flow in the z-plane. On applying the conformal transformation $\zeta = g(z)$, the new plane of flow becomes the ζ -plane. Let ρ be the density of the fluid in both cases. Suppose further that C is a rigid boundary in the z-plane which maps into the curve C' in the ζ -plane. Let the complex velocity potential for the zplane be w = f(z) = ϕ + i ψ where the real functions $\phi(x,y)$, $\psi(x,y)$ are the usual velocity potential and stream function respectively. By means of the transformation $\zeta = g(z)$ it can

express w as a function $\overline{f}(\zeta) = \overline{\phi} + i\overline{\psi}$ where $\overline{\phi} = \overline{\phi}(\xi, \eta)$, $\overline{\psi} = \overline{\psi}(\xi, \eta)$. At the corresponding points t, z, the complex potential w takes the same value so that $\phi = \overline{\phi}, \psi = \overline{\psi}$.

Now C is a rigid boundary in the z-plane and so also a streamline for which $\psi =$ constant. Thus along C', $\overline{\psi} =$ constant. Therefore, C' is a streamline and also a rigid boundary. Therefore, under the conformal transformation, points on the streamline through a given point in the z-plane will transform into points on the stream line through the corresponding point in the ζ -plane. In particular, the boundaries of the fluid in z-plane will transform the boundaries in ζ -plane.

1.2 Transformation of source and sink



Figure 4

Suppose there is a source of strength m at P in the z-plane surrounded by a small closed curve C. By the definition of source, the flow across C is 2π mp. Under the conformal transformation the point P transforms into the point P' in the ζ -plane and the small closed curve C surrounding P in the z-plane transforms into a small closed curve C' surrounding P' in the ζ -plane. The flow across C is given in terms of the stream function by $-\rho \int_{C} d\psi$. Since

each point on C' corresponds to one and only one point on C, this is equal to $-\rho \int_{C'} d\psi$ taken in

the same sense. So, the flow across C' is $2\pi m\rho$ and this will be the same for any small closed curve surrounding P'. Therefore a source transforms into an equal source at the corresponding point. Similarly if there is a sink of strength (-m) at P in the z-plane, then it transforms into an equal sink at the corresponding point in the ζ -plane.

In particular, the boundaries of the fluid in z-plane will transform the boundaries in ζ -plane. And a source, sink or vortex at a particular point in the z- plane will transform an equal source, sink or vortex at the corresponding point in the ζ -plane. The kinetic energy of both corresponding regions is equal.

2. Joukowski Transformation

It is the most common conformal transformation which is given by

$$\zeta = f(z) = z + \frac{a^2}{z}, \qquad (4)$$

where a is constant. The transformation changes z-plane to ζ -plane where z = x + iy. Then,

$$\zeta = \xi + i\eta = z + \frac{a^2}{z},$$

$$= x + iy + \frac{a^{2} (x - iy)}{(x + iy)(x - iy)}.$$

Therefore, $\xi = x (1 + \frac{a^{2}}{x^{2} + y^{2}}), \quad \eta = y(1 - \frac{a^{2}}{x^{2} + y^{2}}).$ (5)
If $x^{2} + y^{2} = r^{2}$, the circle of radius r is in the z-plane, then

$$\frac{\xi^{2}}{\left(r + \frac{a^{2}}{r}\right)^{2}} + \frac{\eta^{2}}{\left(r - \frac{a^{2}}{r}\right)^{2}} = 1$$
, in the ζ -plane. (6)

Therefore by using Joukowski transformation, a circle on the z-plane of radius r transforms into an ellipse with major axis A = $r + \frac{a^2}{r}$ and minor axis B = $r - \frac{a^2}{r}$ on the ζ -plane.



In the special case, when r = a, the ellipse becomes an infinitely thin plate of length 4a ζ -plane, since A = 2a and B = 0. So, the Joukowski transformation changes the circle in the into a flat plate. And then the circle of radius a in the z-plane is called the Joukowski transformation circle.

2.1 Flow Past a Flat Plate with Circulation



Figure 6

The complex potential for a fixed circular cylinder radius a in a stream whose undisturbed spped U makes an angle α with the X-axis and about which there is a circulation κ is

W = U
$$\left(z e^{i\alpha} + \frac{a^2 e^{-i\alpha}}{z} \right) + \frac{i\kappa}{2\pi} \log z$$
. (7)

If the transformation $\zeta = \frac{a^2}{z} + z$ is applied to the whole area outside the circle in the zplane, it transforms into the whole of the ζ -plane with a rigid barrier between the points $(\pm 2a, O)$. The problem then becomes that a flat plate of width 4a, about which there is circulation, in a stream U inclined at α to the plate. Solving for z in terms of ζ ,

$$z = \frac{1}{2} \left\{ \zeta + \sqrt{\zeta^2 - 4a^2} \right\} \text{ and } \frac{a^2}{z} = \frac{1}{2} \left\{ \zeta - \sqrt{\zeta^2 - 4a^2} \right\}.$$

Hence the complex potential (7) becomes

$$\begin{split} \mathbf{W} &= \frac{1}{2} \mathbf{U} \bigg[\left\{ \zeta + \sqrt{\zeta^2 - 4a^2} \right\} e^{i\alpha} + \left\{ \zeta - \sqrt{\zeta^2 - 4a^2} \right\} e^{-i\alpha} \bigg] \\ &+ \frac{i\kappa}{2\pi} \log \frac{1}{2} \bigg\{ \zeta + \sqrt{\zeta^2 - 4a^2} \bigg\} \, . \\ &= \mathbf{U} \bigg[\zeta \cos \alpha + i \bigg\{ \sqrt{\zeta^2 - 4a^2} \bigg\} \sin \alpha \bigg] \\ &+ \frac{i\kappa}{2\pi} \log \frac{1}{2} \bigg\{ \zeta + \sqrt{\zeta^2 - 4a^2} \bigg\} \, , \end{split}$$

neglecting a constant.

The circulation about the plate is given by the decrease in the velocity potential ϕ on describing a circuit round it, this is the same as the decrease in ϕ on describing the corresponding circuit about the cylinder, i.e., there is a circulation κ about the plate.

The velocity at any point can be written

$$-\mathbf{U} + \mathbf{i}\mathbf{V} = \frac{\mathbf{d}\mathbf{W}}{\mathbf{d}\zeta} = \frac{\frac{\mathbf{d}\mathbf{W}}{\mathbf{d}z}}{\left(1 - \frac{\mathbf{a}^2}{z^2}\right)}.$$

The denominator vanishes when $z = \pm a$, i.e., $\zeta = \pm 2a$, therefore the velocity is infinite at both edges unless $\frac{dW}{dz}$ has a factor (z+a) or (z-a), when it will be finite at the corresponding edge.

$$\frac{\mathrm{dW}}{\mathrm{dz}} = \mathrm{U}\left(\mathrm{e}^{\mathrm{i}\alpha} - \frac{\mathrm{a}^2}{\mathrm{z}^2}\mathrm{e}^{-\mathrm{i}\alpha}\right) + \frac{\mathrm{i}\kappa}{2\pi \mathrm{z}},$$

if this is zero when $z = \pm a$, then $\kappa = \pm 4\pi Ua \sin \alpha$. Hence the velocity at the edge $\zeta = 2a$ will be finite.

Example

A flat plate of infinite length and width L is placed in a current of incompressible fluid with its plane at an angle α to the undisturbed stream lines.



Figure 7

The circle of radius a on BOA as diameter transforms into the flat plate B'A' of length 4a, by means of transformation $\zeta = z + \frac{a^2}{z}$. Taking the centre of the circle as origin, the

complex potential in z-plane is given by $W = Uze^{i\alpha} + \frac{Ua^2}{z}e^{-i\alpha} + \frac{i\kappa}{2\pi}\log z$.

$$\frac{\mathrm{dW}}{\mathrm{dz}} = \mathrm{U}\mathrm{e}^{\mathrm{i}\alpha} - \frac{\mathrm{U}\mathrm{a}^2}{\mathrm{z}^2}\mathrm{e}^{-\mathrm{i}\alpha} + \frac{\mathrm{i}\kappa}{2\pi\mathrm{z}}.$$

Stagnation points corresponding to $z = \pm a$ are given by $\frac{dW}{dz} = 0$. Therefore $\kappa = 4\pi a U \sin \alpha$.

By using the Blasiu's theorem,

$$\begin{split} \mathbf{X} - \mathbf{i}\mathbf{Y} &= \frac{1}{2}\mathbf{i}\rho \int \left(\frac{d\mathbf{W}}{d\zeta}\right)^2 d\zeta \,. \\ &= \frac{1}{2}\mathbf{i}\rho \int \left\{ \left(1 + \frac{a^2}{z^2} + \ldots\right) \left(\mathbf{U}e^{\mathbf{i}\alpha} - \frac{\mathbf{U}a^2e^{-\mathbf{i}\alpha}}{z^2} + \frac{\mathbf{i}\kappa}{2\pi z}\right)^2 \right\} dz \,. \end{split}$$

By using Residue Theorem,

$$X - iY = \frac{1}{2}i\rho \frac{2Ui\kappa e^{i\alpha}}{2\pi} 2\pi i = -i\rho\kappa U e^{i\alpha}$$
$$X = \rho\kappa U \sin\alpha = 4\pi\rho a U^2 \sin^2 \alpha$$
$$Y = \rho\kappa U \cos\alpha = 4\pi\rho a U^2 \sin\alpha \cos\alpha.$$

The resultant force R is $4\pi\rho a U^2 \sin \alpha$ and acting at angle, $\tan \theta = \frac{Y}{X}$, $\theta = \frac{\pi}{2} - \alpha$.

N = real part of
$$-\frac{1}{2}\rho \int \left(\frac{dW}{d\zeta}\right)^2 \zeta d\zeta$$

= $-\frac{1}{2}\rho \int \frac{z + \frac{a^2}{z}}{1 - \frac{a^2}{z^2}} \left(\frac{dW}{dz}\right)^2 dz$

$$= \text{real part of} \left[-\frac{1}{2} \rho \left(2U^2 a^2 e^{2i\alpha} - \frac{\kappa^2}{4\pi^2} - 2U^2 a^2 \right) 2\pi i \right]$$
$$= 2\pi \rho U^2 a^2 \sin 2\alpha = \frac{\pi}{8} \rho L^2 U^2 \sin 2\alpha .$$

3. Flow Due to a Source between Two Fixed Boundaries



Figure 8

Consider a source m at the point z_0 in the fluid bounded by the lines $\theta = 0$ and $\theta = \frac{\pi}{3}$. The conformal transformation $Z = z^3$ where $z = re^{i\theta}$ from z-plane transform to Z-plane. The boundaries $\theta = 0$ and $\theta = \frac{\pi}{3}$ in z-plane transform to $\Theta = 0$ and $\Theta = \pi$ (real axis) in Z-plane. The point z_0 in z-plane transforms to point Z_0 in Z-plane such that $Z_0 = z_0^3$ and the source m at z_0 transforms to a source m at Z_0 . Hence the image system with respect to real axis in Zplane consists of a source m at Z_0 and a source m at Z'_0 . Therefore the complex potential for this motion is

$$W = -m \log (Z - Z_0) - m \log (Z - Z'_0)$$

= $-m \log (z^3 - z_0^3) - m \log (z^3 - z'_0^3)$
 $\phi + i\psi = -m \log \{ (z^3 - z_0^3) (z^3 - z'_0^3) \}.$

Example

Suppose that between two fixed boundaries $\theta = \frac{\pi}{4}$ and $\theta = -\frac{\pi}{4}$, there is twodimensional liquid motion due to a source of strength m at the point (a,0) and an equal sink at the point (b,0).



Figure 9

Consider the transformation $Z = z^2$, from xy-plane to $\xi\eta$ -plane, where $z = re^{i\theta}$ and $Z = Re^{i\Theta}$.

Hence the boundaries $\theta = \pm \frac{\pi}{4}$ in the z-plane transform to $\Theta = \pm \frac{\pi}{2}$, the imaginary axis of Z-plane. The points A(a,0) and B(b,0) transform to A'(a²,0) and B'(b²,0) respectively. Since the source transforms to an equal source at A' and the sink transforms to

respectively. Since the source transforms to an equal source at A' and the sink transforms to an equal sink at B', the image system with respective to imaginary axis in Z-plane consists of a source of strength m at $A''(-a^2, 0)$ and a sink of strength -m at $B''(-b^2, 0)$.

Therefore, the complex potential for this motion is

$$W = -m \log(Z - a^{2}) + m \log(Z - b^{2}) - m \log(Z + a^{2}) + m \log(Z + b^{2})$$
$$= -m \log(Z^{2} - a^{4}) + m \log(Z^{2} - b^{4}).$$

By using the transformation,

$$W = -m \log (z^4 - a^4) + m \log (z^4 - b^4)$$
$$= -m \log (r^4 \cos 4\theta - a^4 + ir^4 \sin 4\theta)$$
$$+ m \log (r^4 \cos 4\theta - b^4 + ir^4 \sin 4\theta)$$

$$\psi = -m \left[\tan^{-1} \frac{r^4 \sin 4\theta}{r^4 \cos 4\theta - a^4} - \tan^{-1} \frac{r^4 \sin 4\theta}{r^4 \cos 4\theta - b^4} \right]$$

Thus the stream function of two-dimensional motion due to a source of strength m at (a, 0) and an equal magnitude of sink at (b, 0) is

$$-m\tan^{-1}\frac{r^{4}(a^{4}-b^{4})\sin 4\theta}{r^{8}-r^{4}(a^{4}+b^{4})\cos 4\theta+a^{4}b^{4}}$$
$$\frac{dW}{dz} = -m\frac{4z^{3}}{z^{4}-a^{4}}+m\frac{4z^{3}}{z^{4}-b^{4}}$$

$$=\frac{-4\mathrm{mr}^{3}\left(\cos 3\theta+\mathrm{i}\sin 3\theta\right)\left(\mathrm{a}^{4}-\mathrm{b}^{4}\right)}{\left(\mathrm{r}^{4}\cos 4\theta-\mathrm{a}^{4}+\mathrm{ir}^{4}\sin 4\theta\right)\left(\mathrm{r}^{4}\cos 4\theta-\mathrm{b}^{4}+\mathrm{ir}^{4}\sin 4\theta\right)}$$

The velocity at the point (r, θ) in two-dimensional liquid motion due to a source and sink is

$$q = \left| \frac{dW}{dz} \right| = \frac{4mr^3 (a^4 - b^4)}{(r^8 - 2a^4 r^4 \cos 4\theta + a^8)^{\frac{1}{2}} (r^8 - 2b^4 r^4 \cos 4\theta + b^8)^{\frac{1}{2}}}$$

4. Flow due to a Source and Sink at the Corners of Infinite Rectangle

Consider the infinite rectangle in the z-plane for which $0 \le y \le \pi$, $x \ge 0$. Use the transformation $t = \cosh z$, where $t = \xi + i\eta$ and z = x + iy. Therefore, $\xi = \cosh x \cos y$ and $\eta = \sinh x \sin y$, where $0 \le y \le \pi$, $x \ge 0$. If y = 0 and $0 < x < \infty$, then $1 < \xi < \infty$. If x = 0 and $0 \le y \le \pi$, then $-1 \le \xi \le 1$. If $y = \pi$ and $0 < x < \infty$, then $-\infty < \xi < -1$. Thus, the infinite rectangle in the z-plane for which $0 \le y \le \pi$, $x \ge 0$ into the a half of the t-plane for which η is positive.

Consider the two-dimensional irrotational motion of a liquid due to within the above infinite rectangle with a source and sink are placed at the corners (0,0) and $(0,\pi)$.



Figure 10

The source transforms into an equal source at (1,0) and the sink transforms into an equal sink at (-1,0). The complex potential for this motion is

$$W = -m \log(t-1) + m \log(t+1)$$

= $-m \log \frac{\cosh z - 1}{\cosh z + 1} = -2m \log \tanh \frac{z}{2}$
= $-\lambda \log \tanh \frac{z}{2}$, where $-2m = -\lambda$.
$$\frac{dW}{dz} = -\frac{\lambda}{2} \frac{\operatorname{sec} h^{2} \frac{z}{2}}{\tanh \frac{z}{2}} = -\frac{\lambda}{\sinh z}.$$

For the curve of equal pressure in the liquid, the velocity must be constant. Therefore,

$$q^{2} = \left[-\frac{\lambda}{\sinh z} \right]^{2} = C_{1}^{2}, \text{ where } C_{1} \text{ is a constant.}$$

$$\sinh z \sinh \overline{z} = C_{2}^{2}$$

$$\cosh(z + \overline{z}) - \cosh(z - \overline{z}) = C_{2}^{2}.$$

Therefore, the curves of equal pressure in the liquid are given by $\sinh^2 x + \sin^2 y = C^2$.

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References

[1] Brown, J.W., and Churchill, R.V., "Complex Variables and Applications", Sixth Edition, McGraw-Hill Inc., New York,(1996).

[2] Dr.J.M. Meyers, Dr.D.G.Fletcher. Dr. Y.Dubief., "*Lift and Drag on an Airfoil*", ME 123: Mechanical Engineering Laboratory II : Fluids.

[3] Milne-Thomson, L., "Theoretical Hydrodynamics", Macmillan & Co. Ltd .,London, (1968).

[4] Milne-Thomson, L., . "Theoretical Aerodynamics"., Fourth Edition, Dover Publications, Inc., New York, (1973).

[5] Raisinghania, M. D., "Fluid Dynamics", S. Chand & Co., Ltd., London, Ramagar, New Delhi-110055, (2010)

[6] Wilson, D.H., "Hydrodynamics", Edward Arnold (Publishers) Ltd., London, (1964).